

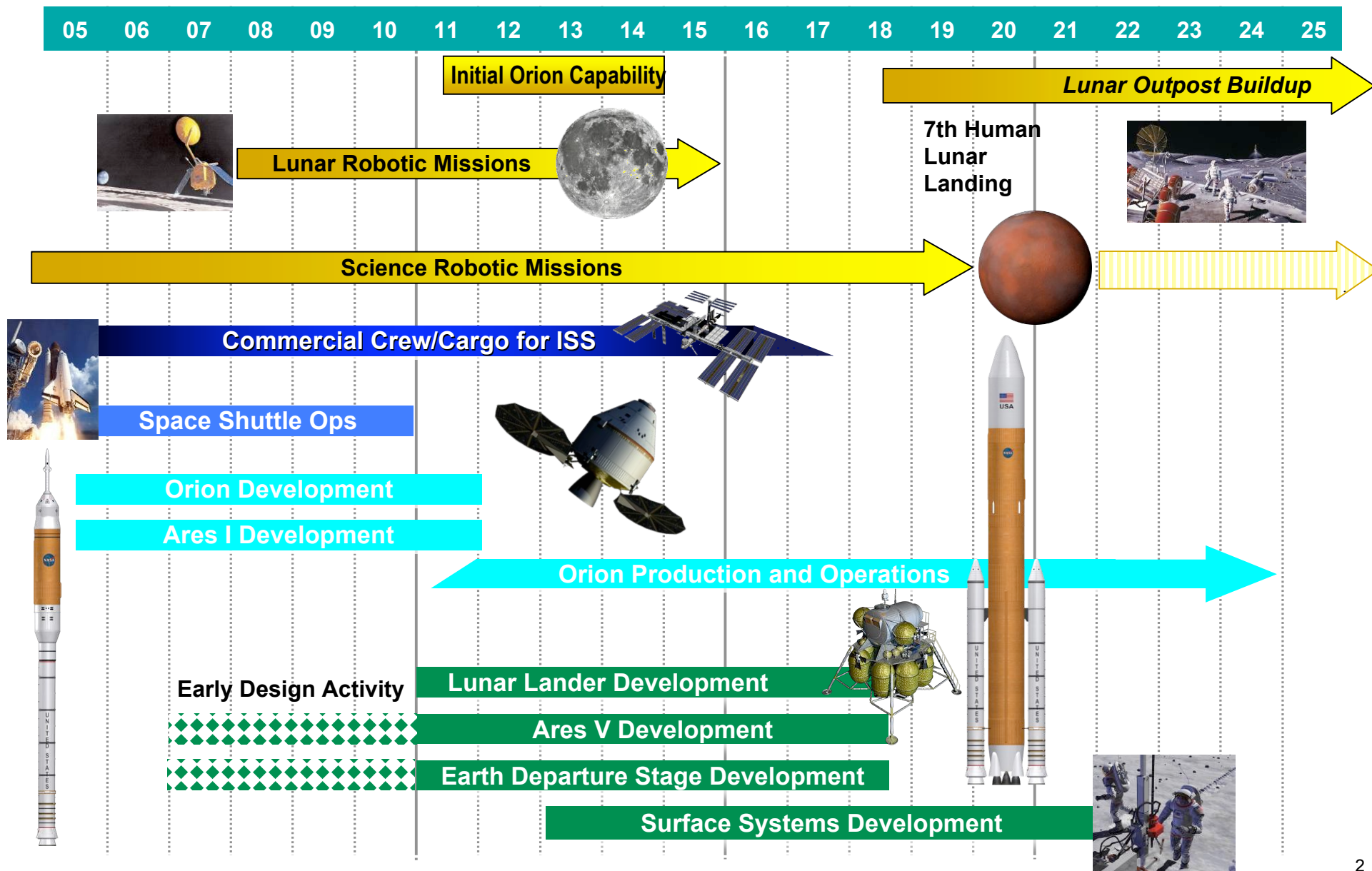


Life Support Technology Challenges for NASA's Constellation Program

Robyn Carrasquillo & Robert Bagdikian
NASA Marshall Space Flight Center
Michael Ewert
NASA Johnson Space Center
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CONSTELLATION

NASA's Exploration Roadmap



Overview of Mission Phases

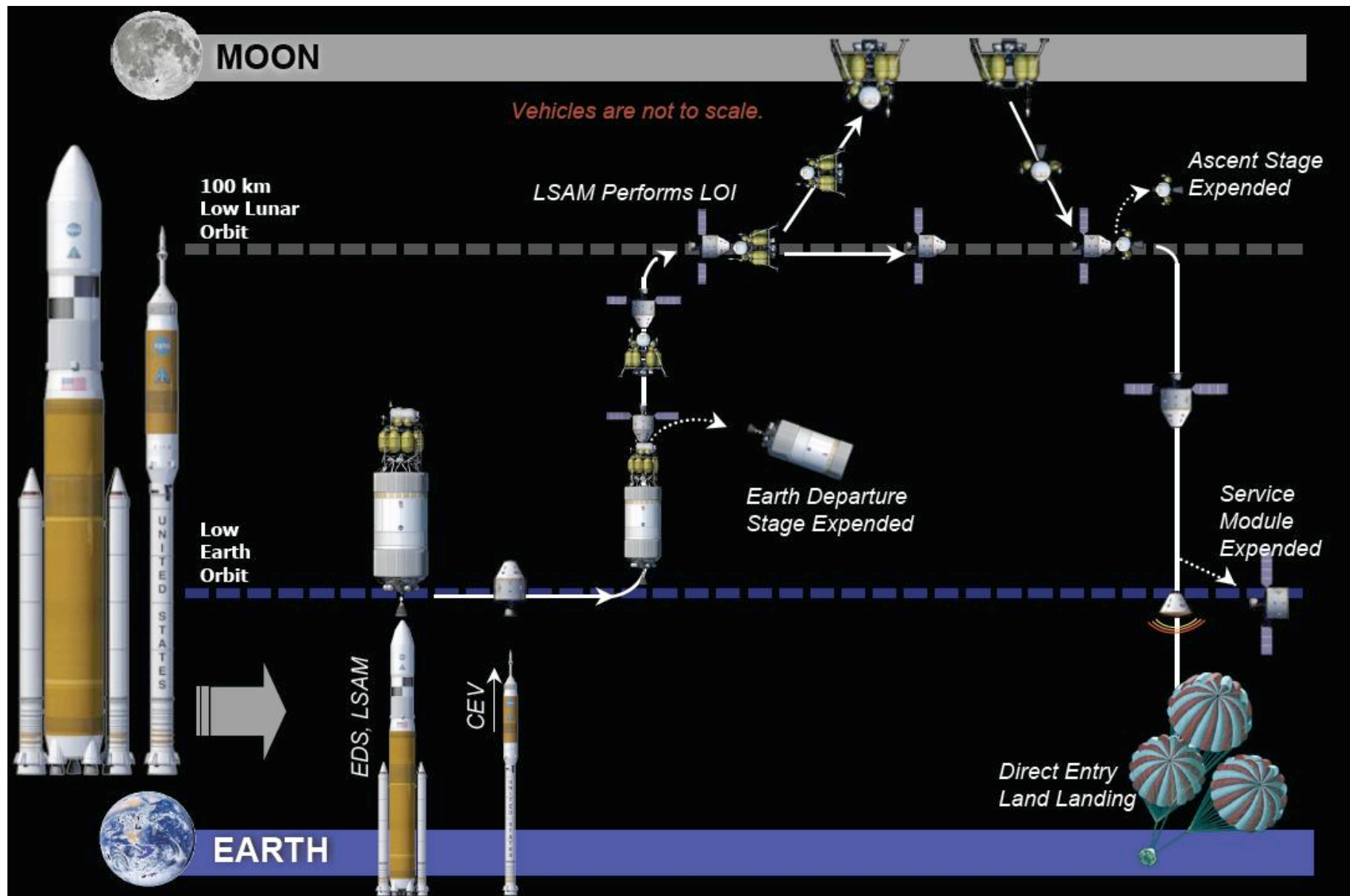
- ◆ **ISS crew/cargo transfer**
- ◆ **Initial Lunar**
- ◆ **Lunar Outpost**
- ◆ **Mars Transit**
- ◆ **Mars Outpost**

ISS Mission

- ◆ **Transport up to 6 crew members on Orion for crew rotation**
- ◆ **210 day stay time**
- ◆ **Emergency lifeboat for entire ISS crew**
- ◆ **Deliver pressurized cargo for ISS resupply**



Lunar Mission



Lunar Lander



- ◆ **Transports 4 crew to and from the surface**
 - Short duration sorties on the surface
 - Lunar outpost crew rotation
- ◆ **Global access capability**
- ◆ **Anytime return to Earth**
- ◆ **Capability to deliver dedicated cargo**
- ◆ **Airlock for surface activities**
- ◆ **Descent stage:**
 - Liquid oxygen / liquid hydrogen propulsion
- ◆ **Ascent stage:**
 - Storable Propellants

Lunar Outpost

- ◆ Habitat supporting 4 crew
- ◆ Capability for daily EVA
- ◆ Use of ISRU
- ◆ Pressurized Rover



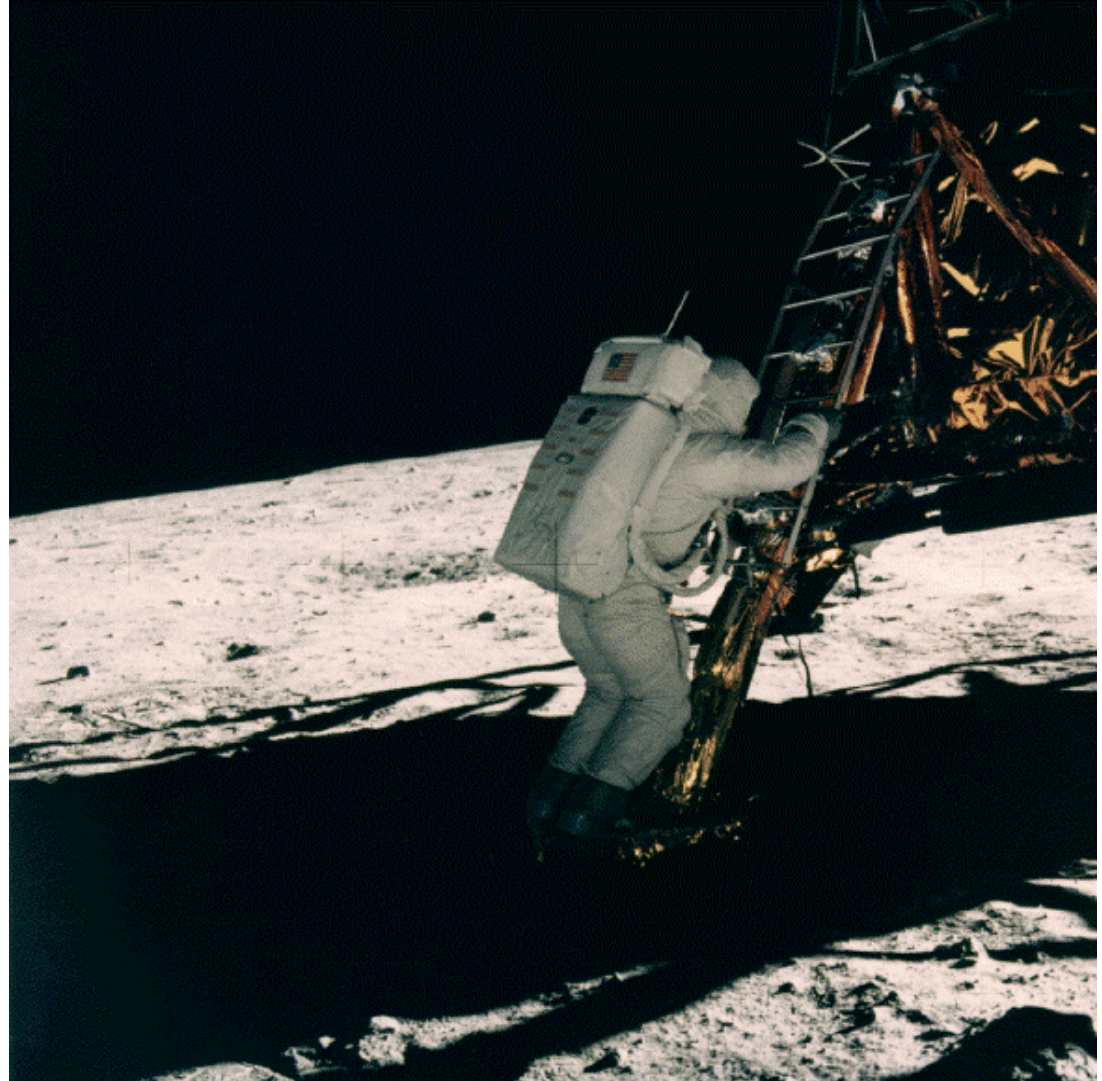
Unique New Requirements and Challenges

- ◆ **Carry up to 6 crew to the ISS**
 - Crew module must support 0-6 crew
- ◆ **Carry 4 crew to the moon**
 - Crew module must orbit moon unmanned for 6 months
- ◆ **Establish permanent Outpost**
 - Cargo lander to leave Outpost building blocks behind
- ◆ **Outpost will have limited resupply capability**
 - Life Support loops must approach closure to minimize resupply needs
 - ~6 months resupply interval
- ◆ **Frequent Outpost EVA's**
 - EVA life support with inherent losses of vented CO₂ and water
- ◆ **Pressurized rover for extended duration EVA's**
- ◆ **132 hour unpressurized survival in Command Module**
 - Life support system must support crew whether in open cabin or suits
- ◆ **Lunar dust environment**
- ◆ **Anywhere access on moon for lunar sorties**

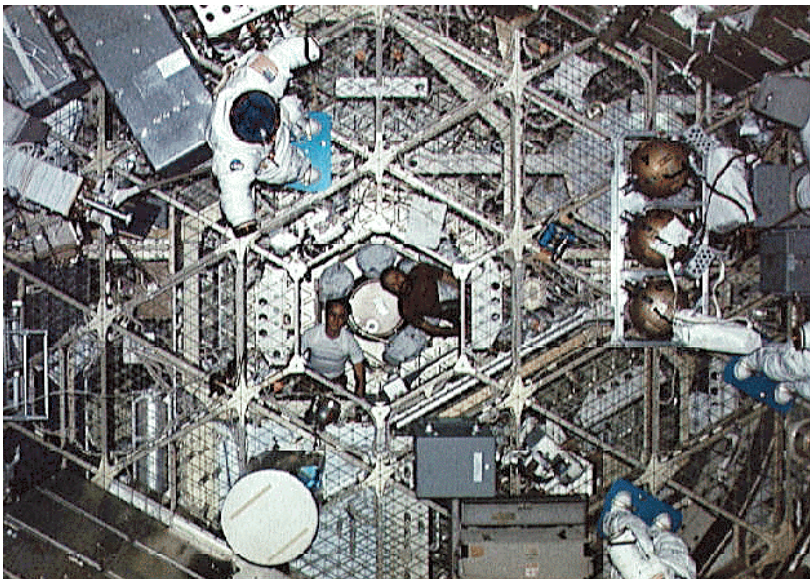
Historical U.S. Life Support Systems

◆ Apollo

- 3 crew
- 6 m³ habitable volume
- 6-12 day mission lengths
- Open loop expendable (LiOH) air revitalization system
- Overboard urine vent
- Rudimentary solid waste collection (bags)
- 100% oxygen environment at 5 psia
- Potable water from fuel cells
- Suit loop for emergency depress survival



Historical U.S. Life Support Systems



◆ Skylab

- 3-person laboratory
- 28-84 day missions
- 361 m³ total habitable volume
- Mixed O₂/N₂ atmosphere at 5 psia; 72% O₂/28% N₂
- 2-bed molecular sieve regenerable CO₂/humidity removal, desorbed to space. Activated charcoal for trace contaminant control during missions; venting of laboratory between missions avoided long-term contaminant buildup.
- Condensing heat exchanger for further humidity control
- Potable water launched with Orbital Workshop
- Disposable bag waste collection

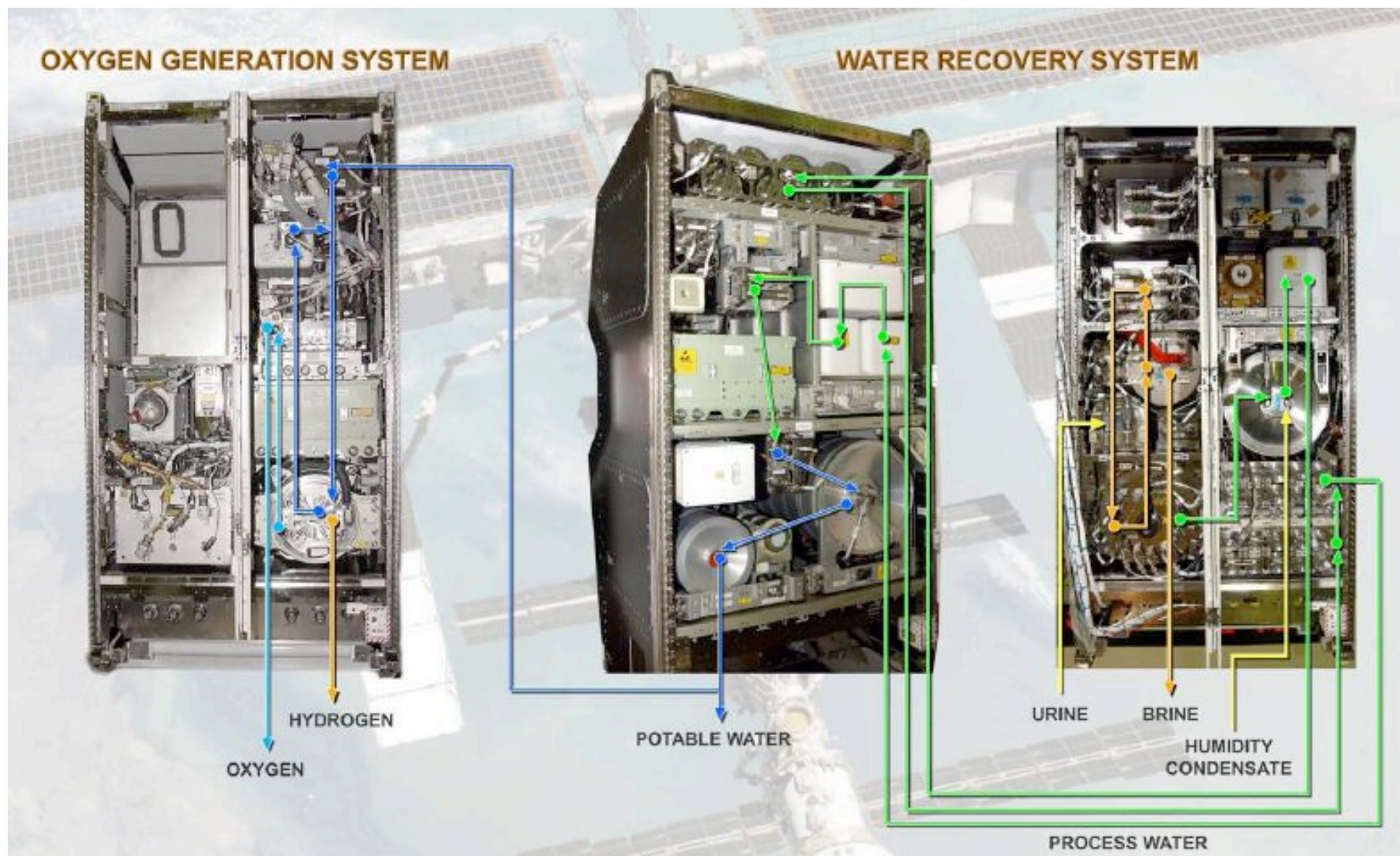
Historical U.S. Life Support Systems

◆ International Space Station

- 3-person crew; to go to 6-crew with activation of Regenerative ECLSS
- Regenerative zeolite CO₂ removal, vented overboard
- Ambient pressure oxygen generation via water electrolysis
- Scar for CO₂ reduction (Sabatier)
- Expendable and catalytic oxidation trace contaminant control
- Urine and humidity condensate water processing (2008)
 - Distillation, multifiltration, and catalytic oxidation
 - 93% recovery of wastewater to potable quality



ISS Regenerative ECLSS



Improvements Needed Over State of the Art Historical Systems – Short-Duration Vehicles

Atmosphere Revitalization

- ◆ **Regenerative open-loop atmosphere revitalization (CEV, lunar sortie lander, pressurized rover)**
 - CO₂ and humidity removal via vacuum swing adsorption eliminates need for condensing heat exchanger and expendable LiOH
 - Recovery of oxygen and water not critical for short-duration missions
 - Candidate technologies include amine and zeolite-based systems
- ◆ **Improved particulate filtration for lunar dust**
 - Filter particles to submicron levels
- ◆ **Emergency breathing mask which does not increase cabin %O₂ to unsafe levels.**
 - Looking at adapting commercial chemical mask.
- ◆ **Targeted trace contaminant adsorbents**
 - Ammonia from amine and suit loop contingency
 - Alcohols typically removed by condensing heat exchanger
- ◆ **Deployable post-fire cleanup device (aka “smoke eater”)**



Improvements Needed Over State of the Art Historical Systems – Short-Duration Vehicles



Atmosphere Monitoring

- ◆ **Post-fire combustion products monitor**
- ◆ **Particulate monitor for lunar dust**
 - Monitor to 0.05 microns
- ◆ **Improved oxygen major constituent monitor**
 - Tighter oxygen control bands require +/- 0.05% accuracy
 - Longer calibration period
- ◆ **Fire detection that eliminates false positive alarms**



Improvements Needed Over State of the Art Historical Systems – Short-Duration Vehicles



Water Storage and Supply

◆ Biocide

- Compatible with materials
- Does not need to be removed for crew health
- Stable for 6 month durations

Waste Collection

◆ Improved urine pretreatment

- Low toxicity, non-corrosive
- Simple introduction method

◆ Solid waste containment that lends itself to transfer to Outpost for water recovery

◆ Simplified, no power, urine separator/vent that works with both genders

- Apollo “can” worked marginally well for males only.
- Needed as backup if spin separator fails

Lunar Surface ECLSS Functions

Pressure Control Subsystem

- ✂ O2 Supply
- ✂ N2 Supply
- ✂ Positive Pressure Relief
- ✂ Intermodule Pressure Equalization
- ✂ Cabin Depress
- ✂ Cabin Pressure Monitoring

Fire Detection & Suppression Subsystem

- ✂ Fire Detection
- ✂ Fire Suppression

Emergency Equipment

- ✂ O2 Masks
- ✂ Toxic Masks

Air Revitalization Subsystem

- ✂ CO2 Removal
- ✂ CO2 Reduction
- ✂ O2 Generation
- ✂ Temperature & Humidity Control
- ✂ Trace Contaminant Control
 - regenerative
 - non-regenerative (for module ingress)
- ✂ Ventilation
 - intramodule
 - intermodule
- ✂ Airborne Particulate Control
- ✂ Atmosphere Composition Monitoring
 - ppO2
 - pp CO2
 - pp H2O (v)
 - Trace Contaminant

Water Recovery & Mgmt Subsystem

- ✂ H2O Recovery
 - Humidity Condensate
 - Waste Hygiene
 - Urine
- ✂ Brine Recovery
- ✂ Water Storage & Distribution
- ✂ Water Quality Monitoring

Waste Mgmt Subsystem

- ✂ Waste Collection & Drying
- ✂ Trash Compaction & Drying

Improvements Needed Over State of the Art Historical Systems – Long-Duration Missions

Atmosphere Revitalization

◆ Atmosphere loop closure

- Improved CO₂ removal – more robust, lower power, integration with CO₂ reduction
 - Structured sorbents to preclude dust generation
 - Water separation which minimizes power/heat for regeneration
 - Mechanical or chemical adsorption-based CO₂ compression and storage
- CO₂ reduction
 - Sabatier only (50% oxygen recovery from CO₂)
 - Complete oxygen recovery from CO₂
 - Challenge is to minimize resupply of catalyst/expendables for this to trade positively over bringing additional water
 - Sabatier plus hydrogen recovery from methane
 - Bosch

Improvements Needed Over State of the Art Historical Systems – Long-Duration Missions

Atmosphere Revitalization, cont.

- ◆ **High pressure oxygen generation for EVA and storage**
 - Possible synergy with Power regenerative fuel cells and ISRU
- ◆ **Potential need for improved hydrogen sensor**
 - Based on ultimate design of HPOGA and other hydrogen-containing systems (like Sabatier, fuel cells, etc).
- ◆ **Improved particulate filtration for lunar dust**
 - Specific application for outpost/airlock
 - Methods to prevent dust from entering airlock
 - Methods to remove dust from atmosphere
 - Robust seals, connectors
- ◆ **Improved Trace Contaminant catalysts, sorbents**
 - Reduce expendables
 - Lower catalytic oxidation temperature
 - Possible photocatalytic filtration of entire air stream to reduce contaminant load of condensate (currently performing trades)
 - Possible incorporation into integrated CO₂ removal/reduction system



Improvements Needed Over State of the Art Historical Systems – Long-Duration Missions



Atmosphere Monitoring

- ◆ **Post-fire cleanup monitor (combustion products) – same as short-duration mission need**
- ◆ **Particulate monitor for lunar dust – same as short-duration mission need**
- ◆ **Trace Contaminant Monitor**
 - Long-duration contaminant buildup concern, and inability to bring back samples for analysis
- ◆ **Microbial monitor**

Improvements Needed Over State of the Art Historical Systems – Long-Duration Missions

Water Processing

◆ Improved water recovery

- >95% water recovery from wastewater (primary processor)
- ~100% water recovery from brine
- Decreased expendables – filters, absorption media
 - Current ISS water recovery system uses 8 lb resupply/100 lb water recovered including maintainable items (2.7 lb expendables/100 lb water recovered)
- Consider use of partial gravity to simplify planetary base system
 - Potential use of modular components that could be added to partial-g system to function in micro-g
- Improved urine pretreat (from short-duration vehicle list) is key to this effort as well.

◆ In-line TOC monitor

- For improved long-term process control and monitoring of water system

◆ Biocide monitor



Improvements Needed Over State of the Art Historical Systems – Long-Duration Missions



Waste Storage/Processing

- ◆ **Recovery of water from solid waste (metabolic and trash)**
 - Tentative target is 50% recovery of water from solid waste
 - Methods to avoid physical transfer of waste from collection container to processor
 - Possibly retrieve containers from Lander to recover resources
- ◆ **Stabilization and long-term storage of solid waste**
 - Could include waste compaction and drying



Improvements Needed Over State of the Art Historical Systems – Long-Duration Missions



Habitability Functions

- ◆ **Laundry**
 - Preliminary trades look favorable
 - Includes need for soap development that is compatible with water processor and crewmembers
 - Lightweight clothing system must also be defined
- ◆ **Vacuum cleaner (for lunar dust) and other crew equipment needs**
- ◆ **Low energy lighting, crew quarters and galley equipment**